Figure 6.24: Unstructured CFD surface grids for the aeroelastic wing deflections across the Mach number regime.

(a) Low subsonic wing flow ($M_s = 0.3$, $\alpha = 2.5^\circ$ altitude = 35,000 ft).

(b) Subcritical wing flow ($M_s = 0.7$, $\alpha = 2.5^\circ$ altitude = 35,000 ft).